

Article

The Stiffness and Damping Characteristics of a Dual-Chamber Air Spring Device Applied to Motion Suppression of Marine Structures

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Abstract: Dual-chamber air springs are used as a key component for vibration isolation in some industrial applications. The working principle of the dual-chamber air spring device as applied to motion suppression of marine structures is similar to that of the traditional air spring, but they differ in their specific characteristics. The stiffness and damping of the dual-chamber air spring device determine the extent of motion suppression. In this article, we investigate the stiffness and damping characteristics of a dual-chamber air spring device applied to marine structure motion suppression using orthogonal analysis and an experimental method. We measure the effects of volume ratio, orifice ratio, excitation amplitude, and frequency on the stiffness and damping of the dual-chamber vibration absorber. Based on the experimental results, a higher-order non-linear regression method is obtained. We achieve a rapid calculation model for dual-chamber air spring stiffness and damping, which can provide guidance to project design.

Keywords: stiffness and damping; orthogonal analysis; vibration suppression; floating marine structure; dual-chamber air spring

1. Introduction

Marine structure safety, especially in the deep sea, is an important concern. The deep-sea floating platform is one of the most important large structures used in ocean energy exploitation. Dynamic responses due to the wind and wave effects must be considered to ensure a safe design [1]. The combined action of complex multiple loads could cause a large vibration amplitude in marine structures [2,3]. Large amplitude and alternating vibration are the main detrimental effects that reduce the safety and fatigue life of a deep-sea platform. In the field of civil engineering, engineers use a wide variety of energy-absorbing devices to reduce the vibration of building structures. These energy-absorbing devices may also be applied to the problem of vibration suppression of deep-sea marine structures.

The vibration of a floating platform could be reduced using active or passive controls [4,5] and various devices have been proposed. A Tuned Liquid Column Damper (TLCD), a kind of passive vibration control device to suppress movement, was proposed by Sakai [6] in 1989. Lee *et al.* [7,8] first used the TLCD to suppress the surge and sway motions vibration of a Tension Leg Platform (TLP). Taflanidis *et al.* [9] further developed a simulation-based method for the design of mass dampers applied for the response mitigation of tension lag platforms. Lee *et al.* [10] experimentally studied the harmonic responses of the TLCD for wind excitations. Tanmoy *et al.* [11] studied the

effective performance of TLCD and another passive vibration-mitigating device, a tuned liquid column ball damper for the control of wave-induced vibration. Zeng *et al.* [12] invented a new type of energy-absorbing device, the S-shaped TLCD. The S-shaped TLCD could effectively suppress the horizontal movement and vertical in-plane rotation of a TLP. Although these devices were effective for the suppression of surge and sway motion, TLCD was ineffective at the suppression of vertical movement, and sometime even enhanced this movement [10]. The vertical movement of the floating structures significantly impacts the strength and fatigue of the tension leg and mooring system, and the inability of these devices to suppress this movement could decrease the safety and service life of the structure.

Vibration-induced vertical motion of TLPs could be balanced by using absorbers with air springs. Dual-chamber air springs are an effective energy-absorbing device used as a key component for vibration isolation. Rijken, Bian, and Spillane *et al.* [13–15] applied a system of vibration absorbers using dual-chamber air springs and water columns to suppress resonant motions and studied the effects of the orifice ratio for structural damping. Bachrach and Rivin [16] studied the complex dynamic stiffness of the damper spring, a function of the excitation frequency. The experimental results measured by Kim and Lee [17] for a dual-chamber pneumatic spring exhibited significant amplitude-dependent nonlinear behavior. Jing *et al.* [18–20] proposed a characteristic output spectrum nonlinear (nCOS) method. The nCOS method is based on the theory of Volterra series expansion. They used the nCOS method for analysis and design of an air spring in a nonlinear vehicle suspension system. These studies focused on traditional forms of a dual-chamber air spring.

A dual-chamber air spring device applied to marine structure motion suppression requires a gas-liquid coupling air spring structure. This differs from the traditional air spring both in structure type and in the vibration characteristics. The compression of the gas in the traditional air spring was controlled by a piston, which acted on the pressure surface of the air spring [21–23]. As shown in Figure 1a, the displacement of the piston, x_0 , causes a compression of the air in the upper chamber. The pressure variation of the air spring chamber is of the same phase and amplitude as the external load. For the gas-liquid coupling air spring device applied to motion suppression of marine structures, a liquid column can also move in an independent manner. As shown in Figure 1b, the amount of compressed air in the air chamber is the relative displacement between the liquid column and the top of the air spring, *i.e.* $x_1 - x_0$. The movement of the liquid column (similar to the piston) and the external excitation can differ in phases and amplitude. The stiffness and damping characteristics of this kind of dual-chamber air spring have not been investigated.

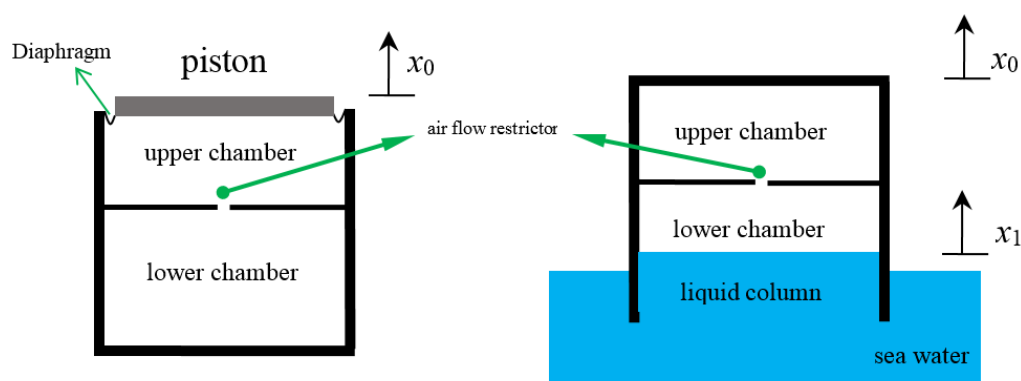


Figure 1. Schematic diagram of two kinds of dual-chamber air spring: (a) the traditional device; (b) a gas-liquid coupling device.

The two main determinants of structure movement suppression effect are the stiffness and damping of the energy-absorbing device. Previous studies of damping and stiffness focused on the traditional air spring, and used an approximation to simulate air flow through the orifice plates [24–26].

These results may not apply to the gas-liquid coupling dual-chamber air spring device. Moreover, few factors were investigated and analyzed. There is not a comprehensive study on which factors affect the stiffness and damping characteristics of dual-chamber air spring. Further studies must be carried out on the gas-liquid coupling dual-chamber air spring device.

Here, we investigate the damping and stiffness of a dual-chamber air spring device applied to motion suppression of marine structures. We used orthogonal analysis and an experimental method and found that stiffness and damping are a function of the volume ratio, orifice ratio, excitation amplitude, and frequency. We then measured the effects of these four factors on dual-chamber vibration absorber stiffness and damping. Based on the experimental results, we used higher-order non-linear regression method and generated a rapid calculation model for dual-chamber air spring stiffness and damping, which can provide guidance for future project design.

2. Theoretical Analysis

The gas-liquid coupling, dual-chamber, air spring energy-absorbing device has two chambers. The upper chamber functions for air storage and has a fixed volume, while the volume of the lower chamber, in contact with the water, varies with the motion of the water column. The gas and the oscillating liquid column are coupled and this device uses the interaction of oscillations of the liquid column and gas to achieve floating structure motion suppression.

The performance and design of the dual-chamber air spring device are usually modeled after a spring damper system. The characteristics of the damping device are described by the spring stiffness and damping coefficient [26–30]. In this work, the dual-chamber air spring device is a complex fluid-gas coupling system, but its dynamic behavior can also be characterized by the spring stiffness and damping coefficient. The physical model of the common spring damping system can be described using a parallel model or as a series model. Here, we explore the advantages and disadvantages of the two models to characterize the dynamic characteristics of the dual-chamber air spring device applied to motion suppression of marine structures.

In this section, three kinds of energy dissipation expression are derived: a series model, a parallel model, and work done by excitation force.

2.1. Parallel Model

As shown in Figure 2, the standard equation of the model is:

$$M\ddot{X} + C\dot{X} + KX = F \quad (1)$$

where M represents the overall mass of the vibration absorber structure, C represents the overall damping of the structure, and K refers to the overall stiffness of the structure.

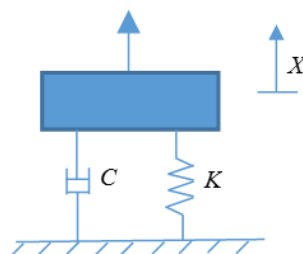


Figure 2. Parallel model.

The structure stiffness in the above equation changes with the air pressure in the vibration absorber air chamber. The change in air pressure is caused by the relative motion of the liquid column in the air chamber. The damping has two components. Some damping is due to fluid vortex shedding at the bottom of the vibration absorber and the friction between the outer wall and water and this

damping is related to the motion of the vibration absorber structure. Damping is also caused by the effect of the air in the air chamber and the orifice as well as the friction between inner wall and water, and this damping is related to the motion of the liquid column in the vibration absorber to the vibration absorber.

For harmonic excitation $F = F_0 e^{i\omega t}$, the structural response X is also a harmonic motion which can be represented as $X = X_0 e^{i(\omega t - \varphi)}$, substituted into Equation (1) and using Euler's Formula, $e^{i\varphi} = \cos\varphi + i\sin\varphi$, the stiffness and damping of the vibration system can be written as:

$$K = \frac{F_0}{X_0} \cos\varphi + \omega^2 M \tag{2}$$

$$C = \frac{1}{\omega} \frac{F_0}{X_0} \sin\varphi \tag{3}$$

The mass matrix can be $M = [m_1, m_2, m_3, m_4, m_5, m_6, m_7]^T$. In the experiment, the mass of each different air chamber length structure is shown as follows:

$$m_1 = 5.769, m_2 = 5.967, m_3 = 6.178, m_4 = 6.386, m_5 = 6.593, m_6 = 6.773, m_7 = 7.001, \text{ in kg.}$$

The periodical energy consumption of damping is:

$$Q = \int C \dot{X} dX = \int_0^T (C \dot{X}) \dot{X} dt = \int_0^T C \dot{X}^2 dt = C \int_0^T \dot{X}^2 dt = \frac{1}{\omega} \frac{F_0}{X_0} \sin\varphi \sum_{i=1}^{i=f_s \cdot T} \dot{X}_i^2 t_i \tag{4}$$

where f_s refers to the sampling frequency of the experiment, $f_s = 1024$ Hz. Due to the equal-interval sampling of the experiment data, $t_i = \frac{1}{1024}$.

2.2. Series Model

As shown in Figure 3, the standard equation of the model is:

$$\begin{cases} M\ddot{X} + k(X - x) = F \\ k(x - X) + c\dot{x} = 0 \end{cases} \tag{5}$$

Substitute $F = F_0 e^{i\omega t}$ and $X = X_0 e^{i(\omega t - \varphi)}$ into the equation and apply Euler's Formula, $e^{i\varphi} = \cos\varphi + i\sin\varphi$, to determine the damping and stiffness:

$$c = \frac{F_0}{\omega X_0} \sin\varphi + \frac{F_0}{\omega X_0} \cot\varphi \cos\varphi + 2\omega M \cot\varphi + \frac{\omega^3 M^2 X_0}{F_0 \sin\varphi} \tag{6}$$

$$k = \frac{c}{\frac{1}{\omega} \cot\varphi + \frac{\omega M X_0}{F_0 \sin\varphi}} \tag{7}$$

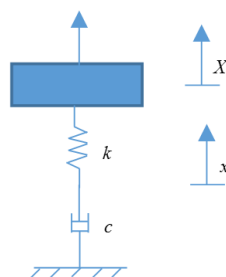


Figure 3. Series model.

The periodical energy consumption of damping is:

$$q = \int c\dot{x}dx = \int_0^T c\dot{x}^2 dt = \sum_{i=1}^{i=f_s \cdot T} c\dot{x}_i^2 t_i \quad (8)$$

2.3. Work Done by Excitation Force

The periodical work done by excitation force of the system is:

$$W = \int FdX = \int_0^T F\dot{X}dt = \sum_{i=1}^{i=f_s \cdot T} F_i \dot{X}_i t_i \quad (9)$$

where F refers to the excitation load, and X refers to the structural response.

3. Experimental Method

3.1. Design of Experimental Device

The dual-chamber air spring vibration absorber described here is mainly used to restrain the large vibration of an offshore structure. In the experiment, a large water container was designed to simulate the action of sea water. The container is a cylindrical device with a diameter of 1500 mm (much larger than the diameter of 100 mm of the air chamber). As shown in Figure 4, the experimental device consists of: 1. Hydraulic pressure loader; 2. Force sensor; 3. Dual-chamber air spring; 4. Orifice plate; and 5. Air pressure sensor.

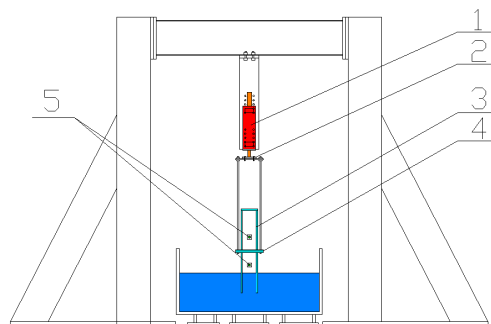


Figure 4. Schematic diagram for the experimental device.

As shown in Figure 5, in order to study the effects of different upper and lower air chamber ratios on the vibrating characteristics, the length of the lower air chamber (200 mm) was kept constant while the upper air chamber was changed (100 mm, 150 mm, 200 mm, 250 mm, 300 mm, 350 mm, 400 mm) to study the effect of different volume ratios. The upper and lower air chambers are connected by an orifice plate and the air flows back and forth through an orifice. The orifice changes the air distribution of the chambers greatly and can influence the vibrating characteristics of the structure. In this experiment, the orifice diameters used to study the effect of different orifice ratios on the air spring were 0, 10, 20, 30, 40, 50, and 60 mm.

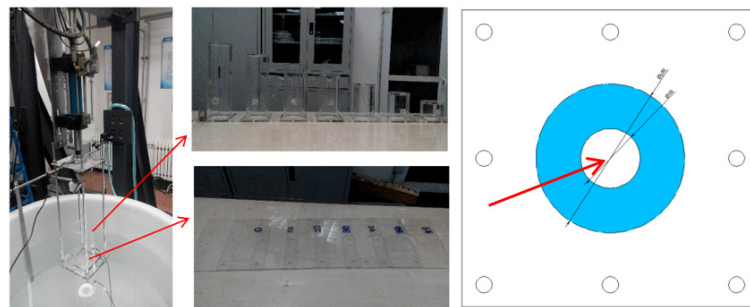


Figure 5. Upper chamber and orifice plates of the dual-chamber air spring.

As shown in Figure 6, a hydraulic servo fatigue machine is used as the loading system of the experiment. The fatigue machine has excellent performance and can be used for large structure experiments. By controlling the loading frequency and amplitude, we can test the effect of external load on the vibrating characteristics of the air spring. In the experiment, the amplitude range was 2, 4, 6, 8, 10, 12, and 14 mm, and the frequency range was 0.497–5.474 Hz.

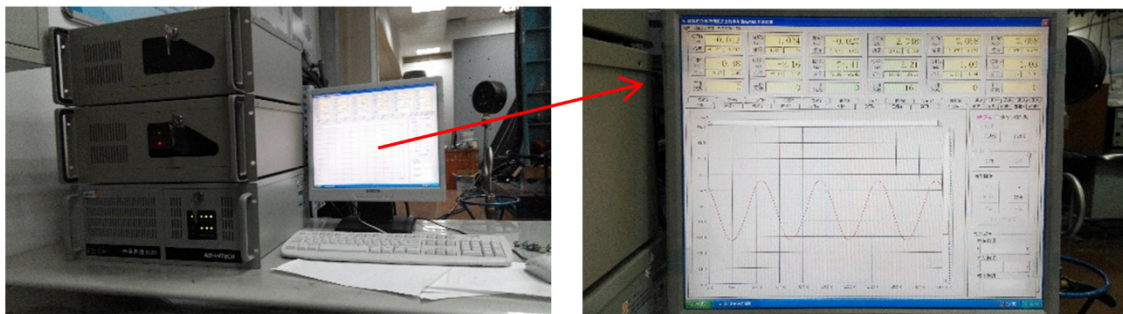


Figure 6. Loading system for experimental device.

3.2. Setting of Test Parameters

We studied the influences of each factor on dual-chamber air spring dynamic stiffness and damping. The experiment assumes the cross section of the upper air chamber equals that of the lower air chamber.

Variables to be considered: Volume of upper air chamber V_2 , Volume of lower air chamber V_1 , opening diameter d , amplitude of outer excitation A , and frequency of outer excitation ω .

The length of the lower air chamber, L_1 , and the frequency of outer excitation, ω , were used as the basic variables with the following three dimensionless numbers:

Height of lower air chamber $L_1 = 0.2$ m

Diameter of air chamber $D = 0.1$ m

$\xi_1 = \frac{L_2}{L_1}$, volume ratio, values = 0.5, 0.75, 1, 1.25, 1.5, 1.75, 2

$\xi_2 = \frac{d}{D}$, orifice ratio, values = 0, 0.1, 0.2, 0.3, 0.4, 0.5, 0.6

$\xi_3 = \frac{A}{L_1}$, ratio of the amplitude of outer excitation and the height of air chamber, values = 0.01, 0.02, 0.03, 0.04, 0.05, 0.06, 0.07.

Frequency of outer excitation. Log value $\ln\omega$ was used to investigate the vibrating characteristics of the air spring at low frequency (values = $-0.7, -0.3, 0.1, 0.5, 0.9, 1.3, 1.7$).

In total, four variables and seven values were tested, as shown in Table 1:

Table 1. Four factors and seven levels chart.

Factor \ Level	Volume Ratio (ξ_1)	Orifice Ratio (ξ_2)	Loading Amplitude Ratio (ξ_3)	Loading Frequency $\ln\omega$	Real Loading Frequency (Hz)
1	0.5	0.0	0.01	-0.7	0.497
2	0.75	0.1	0.02	-0.3	0.741
3	1.0	0.2	0.03	0.1	1.105
4	1.25	0.3	0.04	0.5	1.649
5	1.5	0.4	0.05	0.9	2.460
6	1.75	0.5	0.06	1.3	3.669
7	2.0	0.6	0.07	1.7	5.474

3.3. Orthogonal Experiment and Orthogonal Array

For experimental designs with multiple factors and multiple levels, a large number of experiments are required. For example, for the experimental scheme with four variables and seven levels in this paper, 2401 (7^4) experiments need to be carried out. Hence, a method with the minimum numbers of experiments is needed to find out the effect of each parameter on the vibrating characteristics of the air spring.

In 1947, Rao [31] developed factorial experiments, an optimization experimental method (an efficient testing strategy) that can determine the relation between factors and a test index and specify the primary and secondary factors using fewer trials. Taguchi [32] effectively used orthogonal arrays in his research. Azouzi and Guillot [33] examined the feasibility for an intelligent sensor fusion to estimate online surface finish and dimensional deviations using orthogonal arrays. Green, Krieger, and Wind [34] applied orthogonal arrays to conjoint analysis. The use of the orthogonal test method minimizes the number of tests but still allows determination of the changing rules of all factors. We applied the orthogonal experiment method to investigate the effect of upper and lower air chamber volume ratio, orifice ratio, loading amplitude, and loading frequency of outer excitation.

As an effective method of solving multi-factor experiment problems, orthogonal experimental design selects typical points from the overall experiment and tests them. These typical points are “even” and “regular”: (1) in the experiment, each factor has the same occurrence number at each different level; (2) each combination of two factors at each different level occurred in the experiment and had the same occurrence number. Meeting these two criteria indicates that the experiment scheme designed by orthogonal experiment method is typical. This method reduces the number of experiments and can generally reflect the overall effect of each factor at each level on the index.

Range analysis is commonly used in scientific experiments. Change in experimental results often occurs in scientific experiments due to two types of factors. One is variations arising from random effects; such effects are controllable in experiments, and are therefore inevitable. The other is artificial control produces changes in experimental results. When such factors have a significant impact on the experiment, they are bound to significantly alter the results, accompanied by random factors. Conversely, when such factors have no significant effect on the experimental results, corresponding changes will not be manifested obviously, so changes in experimental results have been substantially ascribed to the effects of random factors. The purpose of conducting scientific experiments is often to determine whether these artificially controllable factors have an impact on the experimental results and what the effect is.

Range analysis is an effective tool to judge the above matters through analyzing the data variation in experimental results, because it can separate random variation from non-random variation in a hybrid state to help determine the source of dominant variations. Using range analysis, we analyzed the results of orthogonal design to explore the effect of various factors on air spring characteristics. In addition, we investigated the effects of four factors (upper and lower air chamber volume ratio, orifice ratio, loading amplitude, and loading frequency of outer excitation) on the air vibration characteristics of the dual-chamber. Seven variation levels of each factor were examined. The tests were performed

49 times, accounting for only 2% of the traditional number of trials. The arrangements and results of the orthogonal analysis scheme are shown in Appendix 1.

4. Experimental Results and Analysis of a Dual-Chamber Air Spring

We used an intuitive method in orthogonal analysis that looks at means and ranges. The mean average is the average value of different factors at the same level, and this can reveal the effect of different levels of single factors on the indicators. The range indicates the maximum value of the average numerical difference for each factor at each level, and reflects the impact of each column factor at different levels, from which the primary and secondary sequence of factors can be determined. If each factor has the same number of level, the influence extent of each factor can be judged directly by comparing the range size.

For 49 sets of experimental schemes, the comparison of energy dissipation is shown in Figure 7.

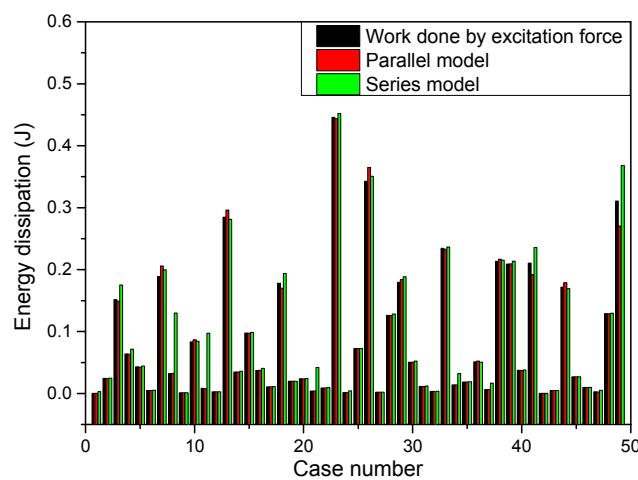


Figure 7. Energy dissipation test program comparison chart.

Figure 7 shows that under most operating conditions, there is a good fit between the parallel model and work done by excitation force, but under individual conditions, the series model shows better goodness of fit. From the relationship of energy dissipation mean of the 49 sets of orthogonal design schemes,

$$\text{the goodness of fit of the parallel model } \alpha_p = 1 - \frac{\sum_0^{49} \frac{|Q_{parallel} - Q_{structure}|}{Q_{structure}}}{49} = 99.88\%.$$

$$\text{The goodness of fit of the series model } \alpha_s = 1 - \frac{\sum_0^{49} \frac{|Q_{series} - Q_{structure}|}{Q_{structure}}}{49} = 89.43\%.$$

By the range relation of energy consumption, we can infer a higher goodness of fit between work done by excitation force and energy dissipation of the parallel model. Furthermore, the parallel model can be employed to make deeper analyses of the structure.

4.1. Influence of Various Parameters on Air Spring Stiffness

Stiffness is a function of the volume ratio, orifice ratio, excitation amplitude, and frequency. The influence law can be seen from the following figures:

As can be seen from Figure 8, the effect of the volume ratio on the dual-chamber vibration absorber stiffness is not monotonic. With an increase in the volume ratio, the stiffness of vibration absorber first slowly decreases and increases at a volume ratio of 1.25, then decreases and finally increases dramatically. As per Figure 9, the effect of the orifice ratio on the dual-chamber vibration absorber

stiffness similarly is not monotonic. With an increase in the orifice ratio, the stiffness of vibration absorber first shows little change, then increases at a orifice ratio of 0.4, then decreases and finally increases to its highest level.

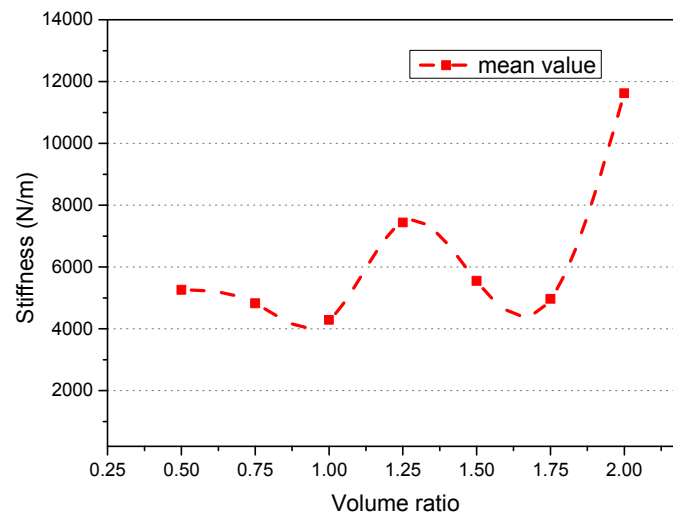


Figure 8. Effect of the volume ratio on the dual-chamber air spring stiffness.

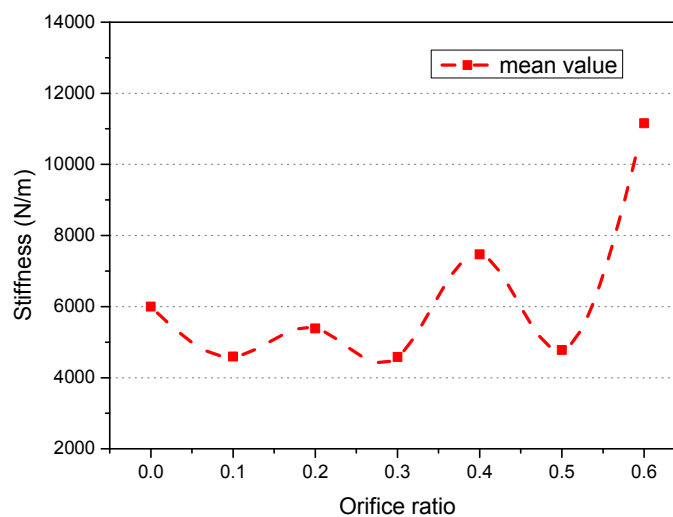


Figure 9. Effect of the orifice ratio on the dual-chamber air spring stiffness.

According to Figure 10, the effect of the loading amplitude ratio on the dual-chamber vibration absorber stiffness is basically monotonic. With an increase in the amplitude ratio, the stiffness of the dual-chamber air spring also increases and slightly protrudes at the amplitude ratio of 0.04. Figure 11 shows the effect of the loading frequency on the dual-chamber vibration absorber stiffness is also monotonic. The stiffness of the dual-chamber air spring monotonically increases with an increase in the loading frequency. As can be seen from Figure 12, the effect of each factor on the dual-chamber vibration absorber stiffness varies, where the volume ratio, orifice ratio, and loading amplitude ratio exert considerable influence but the loading frequency has the maximum impact. The results described in Figure 12 are only applicable when considering the volume ratio, the orifice ratio, load amplitude, and frequency in this case.

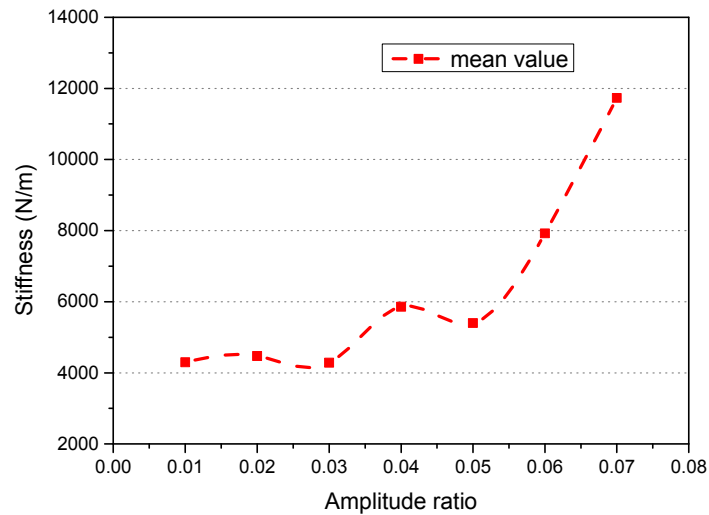


Figure 10. Effect of the amplitude ratio on the dual-chamber air spring stiffness.

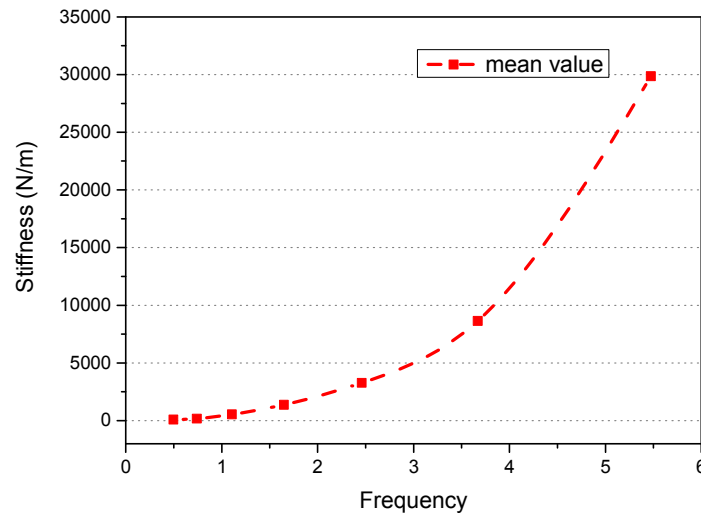


Figure 11. Effect of the frequency on the dual-chamber air spring stiffness.

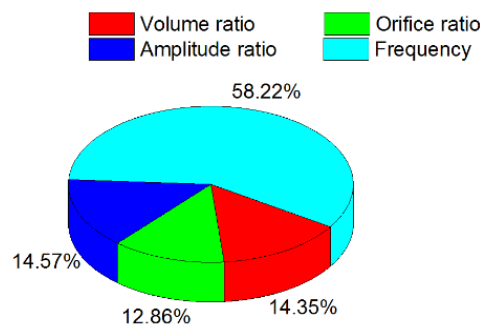


Figure 12. Summary of the effects of the four factors on the dual-chamber air spring stiffness.

4.2. Influence of Various Parameters on Air Spring Damping

Damping is a function of the volume ratio, orifice ratio, excitation amplitude, and frequency. In this section the four factors for the dual-chamber air spring damping effects are examined.

The influence law can be seen from the following figures:

Figure 13 shows that the effect of the volume ratio of the dual-chamber vibration absorber damping is not monotonic. With an increase in the volume ratio, the damping of the vibration absorber first slowly decreases and then increases at a volume ratio of 1.25, then decreases and then increases. As per Figure 14, the effect of the orifice ratio on the dual-chamber vibration absorber damping is also not monotonic. With an increase in the orifice ratio, the damping of vibration absorber first shows little change, then increases at an orifice ratio of 0.4, then decreases and then increases.

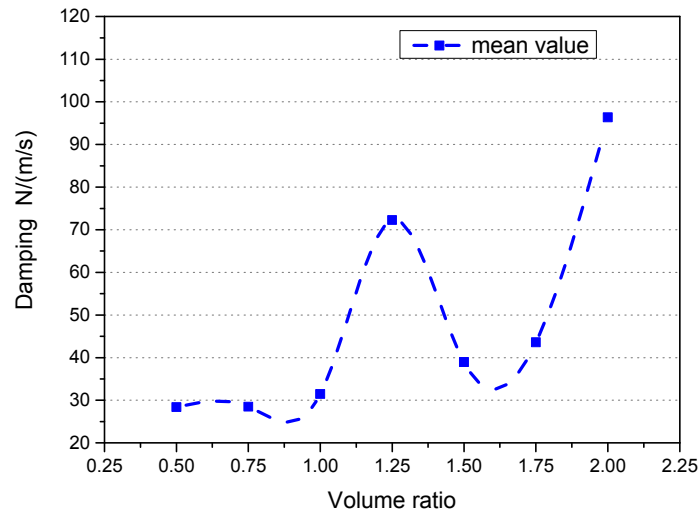


Figure 13. Effect of the volume ratio on the dual-chamber air spring damping.

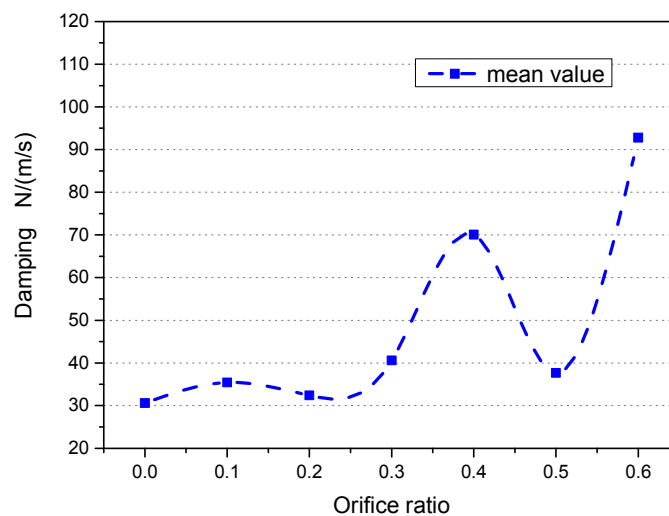


Figure 14. Effect of the orifice ratio on the dual-chamber air spring damping.

According to Figure 15, the effect of the loading amplitude ratio on the dual-chamber vibration absorber damping is not monotonic. With an increase in the amplitude ratio, the structural damping shows little change at first and then declines at 0.05, before eventually increasing. Figure 16 indicates that the effect of the loading frequency on the dual-chamber vibration absorber damping is basically monotonic. With an increase in the amplitude ratio, the vibration absorber damping increases until 3.669 Hz, and then increases dramatically. As summarized in Figure 17, the effect of each factor on the dual-chamber vibration absorber damping varies, where the volume ratio, orifice ratio, and loading amplitude ratio have strong effects but the loading frequency has the maximum impact. The results described in Figure 17 are only applicable when considering the volume ratio, the orifice ratio, load amplitude, and frequency in this case.

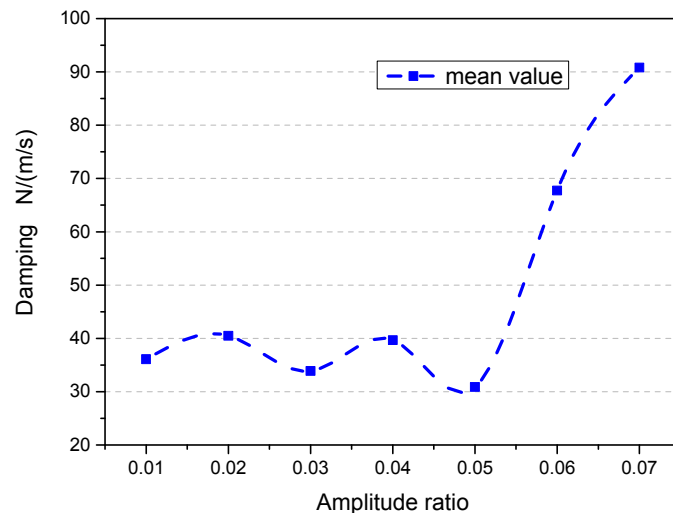


Figure 15. Effect of the amplitude ratio on the dual-chamber air spring damping.

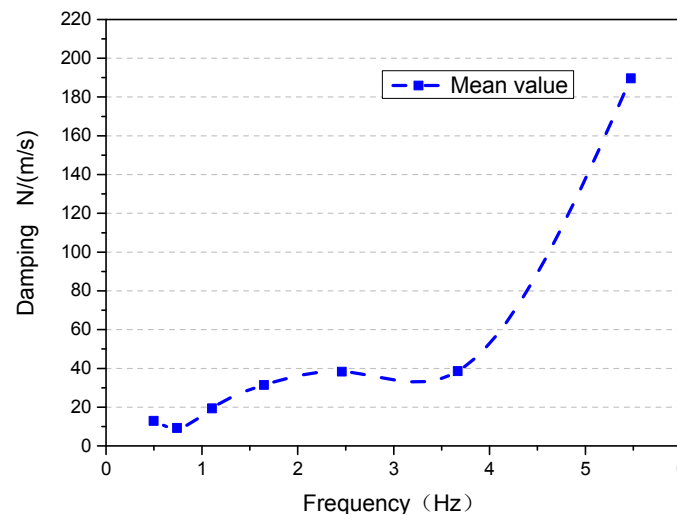


Figure 16. Effect of the frequency on the dual-chamber air spring damping.

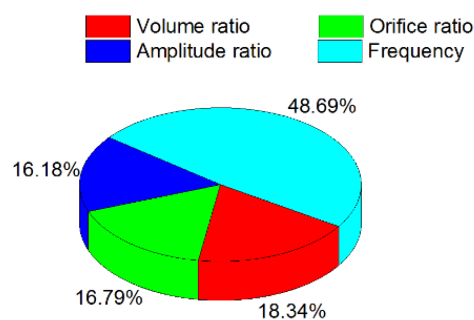


Figure 17. Summary of the effects of the four factors on dual-chamber air spring damping.

Thus, the four factors influence the stiffness and damping of the gas-liquid coupling dual-chamber air spring in a complex manner; there is not a simple linear relationship. Each factor has different effects on stiffness and damping. Prediction of the effects on stiffness and damping of the gas-liquid coupling dual-chamber air spring based on these four factors would be useful for efficient design, and

in the next chapter, we propose an efficient model of dual-chamber air spring stiffness and damping to do this.

5. Rapid Calculation Model of Dual-Chamber Air Spring Stiffness and Damping

From the previous analyses, we can see that volume ratio, orifice ratio, loading amplitude, and loading frequency all have complex effects on the stiffness and damping of the dual-chamber air spring device and are not monotonic in most cases. Therefore complex models must be constructed using statistical analyses of the experimental data.

5.1. Normalization Process of Each Factor

In order to facilitate the analysis of each factor's effect, the normalization process of each factor is performed.

Normalized volume ratio $\varphi_1 = \frac{\xi_1}{2}$, values of 0.25, 0.375, 0.5, 0.625, 0.75, 0.875, 1.

Normalized orifice ratio $\varphi_2 = \frac{\xi_2}{0.6}$, values of 0, 0.167, 0.333, 0.5, 0.667, 0.833, 1.

Normalized loading amplitude ratio $\varphi_3 = \frac{\xi_3}{0.07}$, values of 0.142, 0.286, 0.429, 0.571, 0.714, 0.857, 1.

Normalized loading frequency $\varphi_4 = \frac{\ln\omega + 0.7}{2.4}$, values of 0, 0.167, 0.333, 0.5, 0.667, 0.833, 1.

The functional relationship between the stiffness and damping of the dual-chamber air spring and each factor is:

$$K_{\text{modle}} = f_1(\varphi_1, \varphi_2, \varphi_3, \varphi_4) \tag{10}$$

$$C_{\text{modle}} = f_2(\varphi_1, \varphi_2, \varphi_3, \varphi_4) \tag{11}$$

5.2. Determination of Rapid Calculation Model of Stiffness and Damping

From the above analysis, each factor has a nonlinear impact on the stiffness and damping of the dual-chamber air spring. The quadratic function can not accurately simulate the variation law of the stiffness and damping of the dual-chamber air spring. When the power of an independent variable is more than three, the regression function becomes very unstable [35], thus destabilizing the application of the regression model. In order to more accurately reflect the effect law of each factor on the stiffness and damping of the dual-chamber air spring, this paper uses cubic polynomials, which can usually be converted into ordinary multiple linear regression for processing. Experimental data are used to fit the expression of the stiffness and damping of the dual-chamber air spring.

Suppose the stiffness experimental value meets the following cubic polynomials:

$$\begin{aligned} f_1 &= \beta_0 + \sum_{j=1}^{34} \beta_j g_j(\varphi_1, \varphi_2, \varphi_3, \varphi_4) \\ &= \beta_0 + \beta_1 \varphi_1 + \beta_2 \varphi_2 + \beta_3 \varphi_3 + \beta_4 \varphi_4 + \beta_5 \varphi_1^2 + \beta_6 \varphi_2^2 + \beta_7 \varphi_3^2 + \beta_8 \varphi_4^2 + \beta_9 \varphi_1 \varphi_2 + \beta_{10} \varphi_1 \varphi_3 + \beta_{11} \varphi_1 \varphi_4 \\ &\quad + \beta_{12} \varphi_2 \varphi_3 + \beta_{13} \varphi_2 \varphi_4 + \beta_{14} \varphi_3 \varphi_4 + \beta_{15} \varphi_1^3 + \beta_{16} \varphi_2^3 + \beta_{17} \varphi_3^3 + \beta_{18} \varphi_4^3 + \beta_{19} \varphi_1^2 \varphi_2 + \beta_{20} \varphi_1^2 \varphi_3 + \beta_{21} \varphi_1^2 \varphi_4 \\ &\quad + \beta_{22} \varphi_1 \varphi_2^2 + \beta_{23} \varphi_1 \varphi_3^2 + \beta_{24} \varphi_1 \varphi_4^2 + \beta_{25} \varphi_2^2 \varphi_3 + \beta_{26} \varphi_2^2 \varphi_4 + \beta_{27} \varphi_2 \varphi_3^2 + \beta_{28} \varphi_2 \varphi_4^2 + \beta_{29} \varphi_3^2 \varphi_4 + \beta_{30} \varphi_3 \varphi_4^2 \\ &\quad + \beta_{31} \varphi_1 \varphi_2 \varphi_3 + \beta_{32} \varphi_1 \varphi_2 \varphi_4 + \beta_{33} \varphi_1 \varphi_3 \varphi_4 + \beta_{34} \varphi_2 \varphi_3 \varphi_4 + \varepsilon_i(\sigma^4) \end{aligned} \tag{12}$$

where $\beta_0, \beta_1, \beta_2, \dots, \beta_{34}$ are undetermined coefficients; $\varphi_1, \varphi_2, \varphi_3, \varphi_4$ are controllable variables in the experiment; $g_j(\varphi_1, \varphi_2, \varphi_3, \varphi_4)$ is function expression corresponding to β_j ; and f_1 is the stiffness of dual-chamber air spring.

This paper uses the least squares method to estimate the undetermined coefficients in the above formula to obtain the regression equation for stiffness:

$$\begin{aligned} \frac{\hat{f}_1}{E13} = & -0.0952 + 1.4139\varphi_1 - 0.6852\varphi_2 - 1.3314\varphi_3 + 1.2366\varphi_4 - 0.9296\varphi_1^2 + 0.5123\varphi_2^2 \\ & + 0.3488\varphi_3^2 - 0.7531\varphi_4^2 - 0.3542\varphi_1\varphi_2 - 0.7975\varphi_1\varphi_3 - 1.5502\varphi_1\varphi_4 + 2.7841\varphi_2\varphi_3 \\ & - 0.5386\varphi_2\varphi_4 + 0.2573\varphi_3\varphi_4 - 0.1684\varphi_1^3 - 0.5071\varphi_2^3 + 1.4815\varphi_3^3 - 0.2310\varphi_4^3 \\ & + 0.8761\varphi_1^2\varphi_2 + 1.3092\varphi_1^2\varphi_3 + 0.1184\varphi_1^2\varphi_4 - 0.0996\varphi_1\varphi_2^2 - 0.3868\varphi_1\varphi_3^2 + 0.0888\varphi_1\varphi_4^2 \\ & + 1.1500\varphi_2^2\varphi_3 - 0.6931\varphi_2^2\varphi_4 - 2.5784\varphi_2\varphi_3^2 + 0.1512\varphi_2\varphi_4^2 - 2.7151\varphi_3^2\varphi_4 + 1.6950\varphi_3\varphi_4^2 \\ & - 2.1074\varphi_1\varphi_2\varphi_3 + 1.1258\varphi_1\varphi_2\varphi_4 + 1.3134\varphi_1\varphi_3\varphi_4 + 0.6595\varphi_2\varphi_3\varphi_4 \end{aligned} \tag{13}$$

Similarly, the damping regression equation can be acquired based on the stiffness calculation method:

$$\begin{aligned} \frac{\hat{f}_2}{E11} = & -0.2648 + 3.0819\varphi_1 - 1.6038\varphi_2 - 2.5479\varphi_3 + 2.5604\varphi_4 - 2.1802\varphi_1^2 + 1.4020\varphi_2^2 \\ & + 1.5283\varphi_3^2 - 1.6530\varphi_4^2 - 0.8512\varphi_1\varphi_2 - 1.5508\varphi_1\varphi_3 - 3.7226\varphi_1\varphi_4 + 5.4696\varphi_2\varphi_3 \\ & - 0.8495\varphi_2\varphi_4 + 0.0787\varphi_3\varphi_4 - 0.1734\varphi_1^3 - 0.9562\varphi_2^3 + 1.4337\varphi_3^3 - 0.2797\varphi_4^3 \\ & + 1.6602\varphi_1^2\varphi_2 + 2.1616\varphi_1^2\varphi_3 + 0.8800\varphi_1^2\varphi_4 - 0.1481\varphi_1\varphi_2^2 - 0.3982\varphi_1\varphi_3^2 + 0.6600\varphi_1\varphi_4^2 \\ & + 0.9527\varphi_2^2\varphi_3 - 0.8391\varphi_2^2\varphi_4 - 3.9493\varphi_2\varphi_3^2 + 0.4225\varphi_2\varphi_4^2 - 3.5250\varphi_3^2\varphi_4 + 2.5354\varphi_3\varphi_4^2 \\ & - 3.3547\varphi_1\varphi_2\varphi_3 + 1.6821\varphi_1\varphi_2\varphi_4 + 1.9625\varphi_1\varphi_3\varphi_4 + 0.3758\varphi_2\varphi_3\varphi_4 \end{aligned} \tag{14}$$

Then the stiffness and damping of the dual-chamber air spring can be calculated by the following formulas:

$$K_{\text{modle}} = f_1 (\varphi_1, \varphi_2, \varphi_3, \varphi_4) = \hat{f}_1 \cdot E13 \tag{15}$$

$$C_{\text{modle}} = f_2 (\varphi_1, \varphi_2, \varphi_3, \varphi_4) = \hat{f}_2 \cdot E11 \tag{16}$$

Equations (15) and (16) determine the cubic polynomial equation of stiffness and damping.

5.3. Test of Calculation Model

In statistics, regression analysis is performed using the variables. The coefficient of determination, R^2 is the ratio between the regression sum of squares and sum of squares for total. R^2 is between 0 and 1. The closer it is to 1, the better the fit of the regression. A goodness of fit of more than 0.8 is typically considered acceptable.

In multiple regression analysis, R^2 expression:

$$R^2 = 1 - \frac{SSE}{SST} \tag{17}$$

where SST is the sum of squares for total; and SSE is the sum of squares for error.

$$SST = \sum_{i=1}^N (f_{1i} - \bar{f}_{1i})^2 \tag{18}$$

$$SSE = \sum_{i=1}^N (f_{1i} - \hat{f}_{1i})^2 \tag{19}$$

The coefficient of determination R^2 is related to the number of independent variables. In order to accurately test the accuracy of the model, it is necessary to take into account the degrees of freedom and make adjustments of R^2 expression according to the size of the formula.

$$R_a^2 = 1 - \frac{SSE/(n-p-1)}{SST/(n-p)} = 1 - (1 - R^2) \cdot \left(\frac{n-1}{n-p-1} \right) \tag{20}$$

R_a^2 is denoted as the adjusted coefficient of determination; n is the total number of experiments, and p is the number of variables. The adjusted coefficient of determination more accurately reflects the degree of fit between the model and experimental data.

By making an analysis of the data ($n = 49, p = 35$), the following can be obtained:

1. For the stiffness model $R^2 = 0.99154, R_a^2 = 0.96378$, the goodness of fit is 96.378%, indicating that the model can well simulate and predict the experimental results. This indicates the effectiveness of the cubic polynomial regression equation fit.
2. For the damping model, $R^2 = 0.98970, R_a^2 = 0.95470$, the goodness of fit is 95.470%, again indicating that the model can well simulate and predict the experimental results. This again supports the effectiveness of the cubic polynomial regression equation fit.

5.4. Prediction of Experimental Results by the Rapid Calculation Model

To further validate the forecast accuracy of the stiffness and damping rapid calculation model, we conducted a series of experiments. To ensure the universality, these experiments consist of random combinations of four factors at different levels, different from the levels tested in the test programs in the orthogonal table.

The straight lines in Figures 18 and 19 are experimental values, while red dots are predictive values for the rapid calculation model. If a dot falls on the line, this indicates the predicted value is equivalent to the experimental value; deviations from the straight line indicate differences between the predicted and the experimental values. As seen in the figures, the predicted and experimental values fit well. For more horizontal level combination schemes, the predicted value can be calculated based on the rapid calculation model proposed in this paper, which can play a guiding role to engineering structural design and applications.

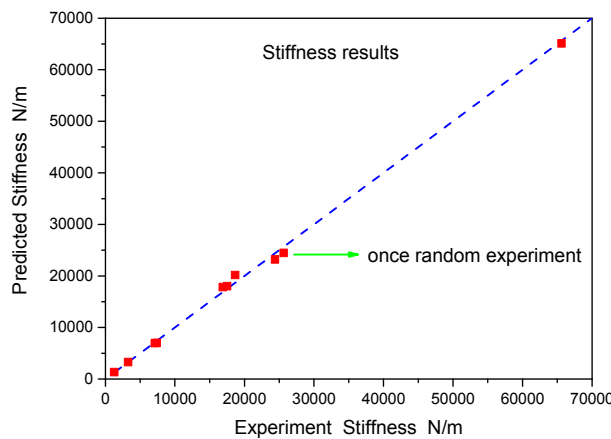


Figure 18. Comparison of experimental and predicted values on the dual-chamber air spring stiffness.

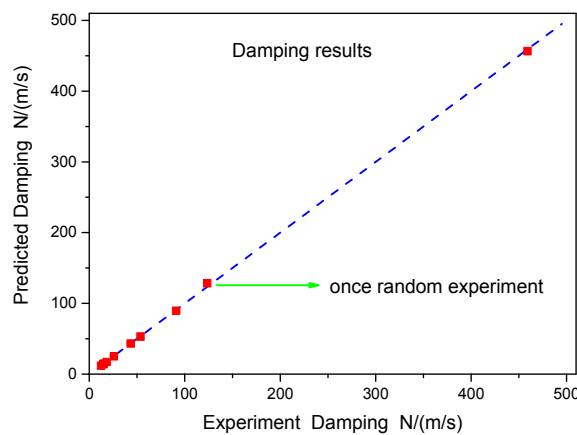


Figure 19. Comparison of experimental and predicted values on the dual-chamber air spring damping.

The experiments performed here were representative in the parameter range, and the polynomial prediction model is appropriate for these values. The application of the polynomial prediction model is as follows: upper and lower air chamber volume ratio, $\xi_1 \in [0.5 - 2.0]$, orifice ratio, $\xi_2 \in [0 - 0.6]$, loading amplitude ratio, $\xi_3 \in [0.01 - 0.07]$, and loading frequency $f \in [0.497 - 5.474]$.

6. Conclusions

The mechanical differences between the gas-liquid coupling dual-chamber air spring device and the traditional one make the analysis of effects on stiffness and damping more challenging. The compression of the gas in the traditional air spring is directly on the external load of the air chamber by the piston. The pressure variation of the air spring chamber is the same phase and amplitude as the external load, but for the dual-chamber air spring device applied to motion suppression of marine structures, the movement of the liquid column and the external excitation differ in phases and amplitude.

The current study analyzes the characteristics of a dual-chamber air spring device. A parallel model and series model were used to simulate the dual-chamber air spring energy-absorbing device. An orthogonal test scheme was used to investigate the effect laws of four factors (upper and lower air chamber volume ratio, orifice ratio, loading amplitude, and loading frequency of outer excitation) on the air vibration characteristics of the dual-chamber. Based on the experimental results, a higher-order non-linear regression method was obtained, achieving a rapid calculation model for dual-chamber air spring stiffness and damping, and the reliability of the method was verified experimentally. Using the rapid calculation model, from the upper and lower air chamber volume, orifice ratio, the frequency, and amplitude of external load, we can determine the stiffness and damping of the dual-chamber air spring device applied to motion suppression of marine structures by the formulas (15) and (16). The main findings are:

1. Based on energy consumption results, the goodness of fit of the parallel model was 89.43%, and the goodness of fit of the series model was 99.88%. The parallel model is more consistent with the real physical model.
2. The effects of volume ratio and orifice ratio on dual-chamber vibration absorber stiffness were not monotonic, but the loading amplitude ratio and frequency tended toward monotonic increasing.
3. The effects of the volume ratio, orifice ratio, and loading amplitude ratio on the dual-chamber vibration absorber stiffness did not behave in a monotonic manner, but the loading frequency on damping tended toward monotonic increasing.
4. A polynomial rapid calculation model for stiffness and damping was constructed. The accuracy of the rapid calculation model results was verified by the experimental results, and the predicted values were in good agreement with the experimental values.

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Conflicts of Interest: The authors declare no conflict of interest.

Appendix

Table A1. Orthogonal table L49 and the experimental analysis results.

Case	Volume Ratio (ξ_1)	Orifice Ratio (ξ_2)	Loading Amplitude Ratio (ξ_3)	Loading Frequency $\ln\omega$	Real Loading Frequency (Hz)	Parallel Model			Series Model			Work Done by Excitation Force
						K	C	Q	k	c	q	W
1	0.5	0	0.01	-0.7	0.497	109.192	14.772	0.0004292	128.627	97.766	0.003081	0.0004297
2	0.5	0.1	0.03	0.5	1.649	1249.035	19.870	0.02411	1282.923	752.237	0.02483	0.02437
3	0.5	0.2	0.05	1.7	5.474	24512.047	58.106	0.1492	24674.823	8808.250	0.1752	0.1516
4	0.5	0.3	0.07	0.1	1.105	503.285	15.400	0.06352	525.984	356.854	0.07156	0.06384
5	0.5	0.4	0.02	1.3	3.669	7208.513	47.043	0.04246	7371.526	2127.292	0.04434	0.04304
6	0.5	0.5	0.04	-0.3	0.741	158.223	7.249	0.005029	165.412	166.801	0.005102	0.005048
7	0.5	0.6	0.06	0.9	2.460	3121.949	36.450	0.2058	3223.487	1157.179	0.1996	0.1887
8	0.75	0	0.07	1.3	3.669	10804.935	10.771	0.03249	10810.636	20423.049	0.1299	0.03209
9	0.75	0.1	0.02	-0.3	0.741	152.925	5.989	0.001248	158.002	186.403	0.001331	0.001287
10	0.75	0.2	0.04	0.9	2.460	3298.782	35.056	0.08677	3387.664	1336.123	0.08447	0.08329
11	0.75	0.3	0.06	-0.7	0.497	98.908	8.265	0.008236	105.624	129.970	0.09743	0.008478
12	0.75	0.4	0.01	0.5	1.649	1277.190	20.470	0.002832	1312.362	763.784	0.002948	0.002881
13	0.75	0.5	0.03	1.7	5.474	17619.902	102.952	0.2962	18330.771	2654.775	0.2812	0.2846
14	0.75	0.6	0.05	0.1	1.105	519.751	15.930	0.03474	543.270	367.970	0.03593	0.03468
15	1	0	0.06	0.5	1.649	1412.352	24.368	0.09738	1457.423	787.955	0.09871	0.09771
16	1	0.1	0.01	1.7	5.474	17009.600	98.264	0.03743	17680.432	2589.838	0.04081	0.0374
17	1	0.2	0.03	0.1	1.105	561.650	16.714	0.01086	585.609	408.524	0.01111	0.01092
18	1	0.3	0.05	1.3	3.669	7642.988	36.205	0.17	7734.053	3074.820	0.1937	0.178
19	1	0.4	0.07	-0.3	0.741	179.227	8.004	0.02	186.963	193.429	0.02007	0.02
20	1	0.5	0.02	0.9	2.460	3121.603	28.248	0.02371	3182.590	1474.095	0.02413	0.02384
21	1	0.6	0.04	-0.7	0.497	99.429	8.566	0.004021	106.606	127.237	0.04211	0.004046
22	1.25	0	0.05	-0.3	0.741	175.538	7.100	0.00899	181.754	207.608	0.009773	0.009058
23	1.25	0.1	0.07	0.9	2.460	3294.971	45.801	0.4436	3446.867	1039.328	0.452	0.446
24	1.25	0.2	0.02	-0.7	0.497	93.417	19.146	0.001581	131.580	66.013	0.004269	0.001549
25	1.25	0.3	0.04	0.5	1.649	1337.610	35.092	0.07292	1436.307	510.685	0.07293	0.07287
26	1.25	0.4	0.06	1.7	5.474	39581.430	324.946	0.365	42733.898	4404.865	0.3508	0.3427

Table A1. Cont.

Case	Volume Ratio (ξ_1)	Orifice Ratio (ξ_2)	Loading Amplitude Ratio (ξ_3)	Loading Frequency $\ln\omega$	Real Loading Frequency (Hz)	Parallel Model			Series Model			Work Done by Excitation Force
						K	C	Q	k	c	q	W
27	1.25	0.5	0.01	0.1	1.105	507.804	26.241	0.002135	573.125	230.241	0.002108	0.002117
28	1.25	0.6	0.03	1.3	3.669	7085.175	47.500	0.1258	7254.263	2037.839	0.1285	0.1262
29	1.5	0	0.04	1.7	5.474	25749.741	114.565	0.184	26352.096	5012.048	0.1885	0.1795
30	1.5	0.1	0.06	0.1	1.105	578.651	17.741	0.05048	604.853	409.543	0.05246	0.05053
31	1.5	0.2	0.01	1.3	3.669	7492.063	34.781	0.01139	7577.801	3074.069	0.01196	0.0116
32	1.5	0.3	0.03	-0.3	0.741	191.848	8.807	0.003483	200.600	201.881	0.003591	0.003496
33	1.5	0.4	0.05	0.9	2.460	3380.530	47.534	0.2329	3540.001	1055.189	0.2364	0.2345
34	1.5	0.5	0.07	-0.7	0.497	83.222	14.337	0.01418	107.242	64.009	0.03212	0.01412
35	1.5	0.6	0.02	0.5	1.649	1376.694	34.954	0.01857	1471.837	540.732	0.01881	0.01865
36	1.75	0	0.03	0.9	2.460	3187.457	27.938	0.05243	3245.880	1552.170	0.05043	0.05119
37	1.75	0.1	0.05	-0.7	0.497	85.259	12.130	0.006464	102.043	73.747	0.01677	0.006484
38	1.75	0.2	0.07	0.5	1.649	1529.732	46.205	0.2168	1679.347	518.626	0.2152	0.2133
39	1.75	0.3	0.02	1.7	5.474	18778.734	133.334	0.2094	19897.493	2371.391	0.2137	0.2089
40	1.75	0.4	0.04	0.1	1.105	573.199	29.053	0.03713	644.131	263.825	0.03814	0.03742
41	1.75	0.5	0.06	1.3	3.669	10461.160	45.520	0.1921	10566.332	4573.206	0.2358	0.2106
42	1.75	0.6	0.01	-0.3	0.741	174.026	10.879	0.0004944	188.745	139.501	0.0004885	0.0004824
43	2	0	0.02	0.1	1.105	564.645	14.696	0.005067	583.071	465.062	0.005049	0.005031
44	2	0.1	0.04	1.3	3.669	9798.212	48.073	0.179	9923.452	3809.107	0.1693	0.1718
45	2	0.2	0.06	-0.3	0.741	198.357	16.858	0.02693	229.366	124.692	0.02689	0.02692
46	2	0.3	0.01	0.9	2.460	3509.969	47.140	0.009806	3661.021	1142.528	0.009924	0.009847
47	2	0.4	0.03	-0.7	0.497	83.851	13.458	0.002517	104.857	67.178	0.005397	0.002638
48	2	0.5	0.05	0.5	1.649	1490.134	39.147	0.1286	1600.387	568.245	0.1297	0.129
49	2	0.6	0.07	1.7	5.474	65716.150	495.144	0.2703	70124.859	7875.753	0.368	0.3109

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